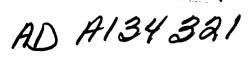




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REPORT NO. NADC-81118-60



HELICOPTER FLYING QUALITIES CHARACTERISTICS-CH-46E VOLUME 2

BOEING VERTOL CO. Boeing Center, P.O. Box 16858 Philadelphia, PA 19142

3 OCTOBER 1983

FINAL REPORT
Program Element No. 62241N
Work Unit No. 41400



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Prepared For
Aircraft & Crew Systems Technology Directorate (Code 6053)
NAVAL AIR DEVELOPMENT CENTER
Warminster, PA 18974

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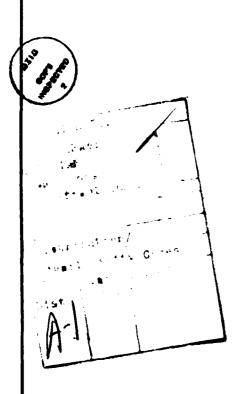
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20 ABSTRACT COMMENT OF THE STATE OF THE STAT	black by black and an			
This document defines the flying qualities characteristics of the CH-46E helicopter. The data are representative of both the				
metal-bladed and composite-bladed versions. Analytically com-				
puted static trim data are presented for a wide range of con-				
figurations (gross weight, c.g.) and flight conditions (airspeed, altitude, sideslip, climb, autorotation). Correlation of trim				
altitude, sideslip, climb, a	utorotation).	provided for validation		
data with available flight test data is provided for validation. (see next page)				
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20. ABSTRACT (continued)

Analytically computed static stability and control derivatives are compiled for significant combinations of configuration and flight condition. Time history data relating to dynamic stability, control response and SAS failures are extracted from flight test records obtained during the Contractor's CH-46E SLEP II flight test program.



VOLUME 2

CH-46E TRIM CHARACTERISTICS

This Volume contains plotted trim data obtained from the Boeing Vertol Tandem Rotor Trim and Stability Analysis Program (A-97) for the CH-46E helicopter.

A catalog of the flight conditions at which the trim characteristics were determined is given on Page 2. Data for the following flight regimes are presented.

- Level flight; from 40 knots rearward to envelope-limited maximum forward speed, at sea level and at service ceiling for the two gross weights considered.
 - Maximum power climb and autorotation from 60 knots to 120 knots at sea level, and from 70 knots to enveloped limited maximum speed at service ceiling, for two gross weights.
 - Constant altitude sideslips at sea level, to ±45 deg at 50 knots, ±30 deg at intermediate speed, and to power-limited sideslip angle at high speed, for two gross weights.

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Sideward flight to 45 knots left and right at sea level, for two gross weights.

For each flight regime considered, the four cockpit control positions, plus helicoter pitch and roll attitudes, are plotted vs. airspeed or sideslip angle.

-1

CATALOG OF TRIM FLIGHT CONDITIONS A97 TRIM ANALYSIS FOR CH-46E

Figure 1	Level Flight	Page 3
	24300 lb., 6 in. fwd, 0 ft.	
	24300 lb., 13 in. fwd, 0 ft.	
	24300 lb., 6 in. fwd, 8000 ft.	
Figure 2	Level Flight	Page 6
	17500 lb., 20 in. aft, 0 ft.	
	17500 lb., 40 in. fwd, 0 ft.	
	17500 lb., 20 in. aft, 14000 ft.	
Figure 3	Max. Power Climb and Autorotation	Page 9
	24300 lb., 6 in. fwd, 0 ft.	
	24300 lb., 6 in. fwd, 8000 ft.	
Figure 4	Max. Power Climb and Autorotation	Page 12
	17500 lb., 20 in. aft, 0 ft.	
	17500 lb., 20 in. aft, 14000 ft.	
Figure 5	Sideslip	Page 15
	24300 lb., 6 in. fwd, 0 ft., 50 kt	
	24300 lb., 6 in. fwd, 0 ft., 90 kt	
	24300 lb., 6 in. fwd, 0 ft., 130 kt	
Figure 6	Sideslip	Page 18
	17500 lb., 20 in. aft, 0 ft., 50 kt	
	17500 lb., 20 in. aft, 0 ft., 95 kt	
	17500 lb., 20 in. aft, 0 ft., 140 kt	
Figure 7	Sideward Flight	Page 21
	24300 lb., 6 in. fwd, 0 ft.	
	17500 lb 20 in aft 0 ft	

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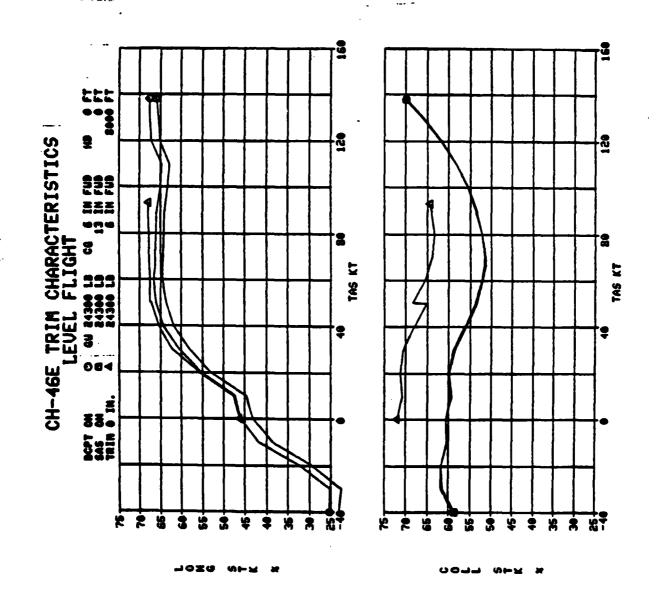


FIGURE 1A

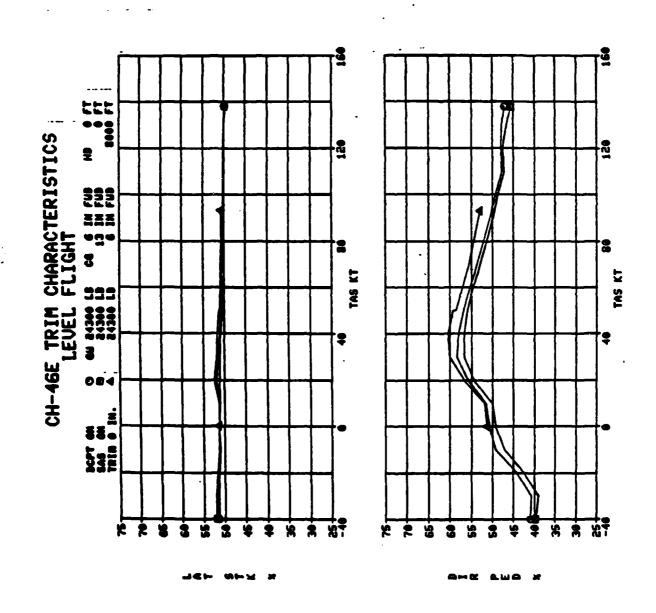


FIGURE 1B

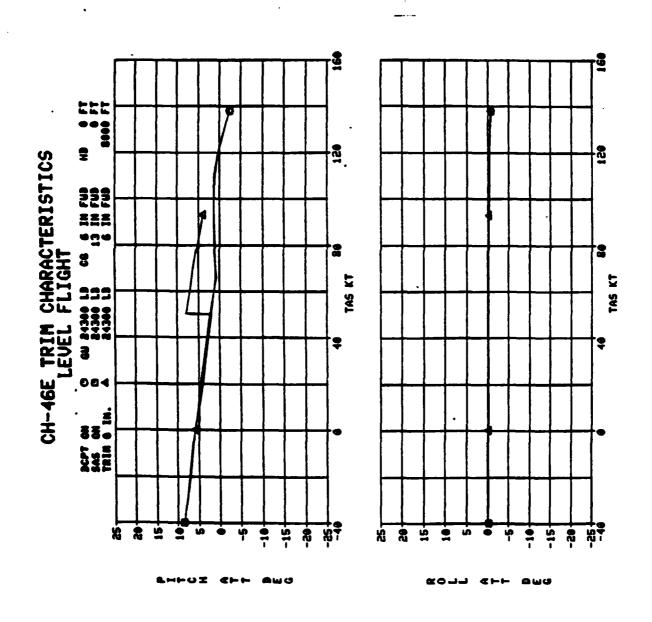


FIGURE 1C

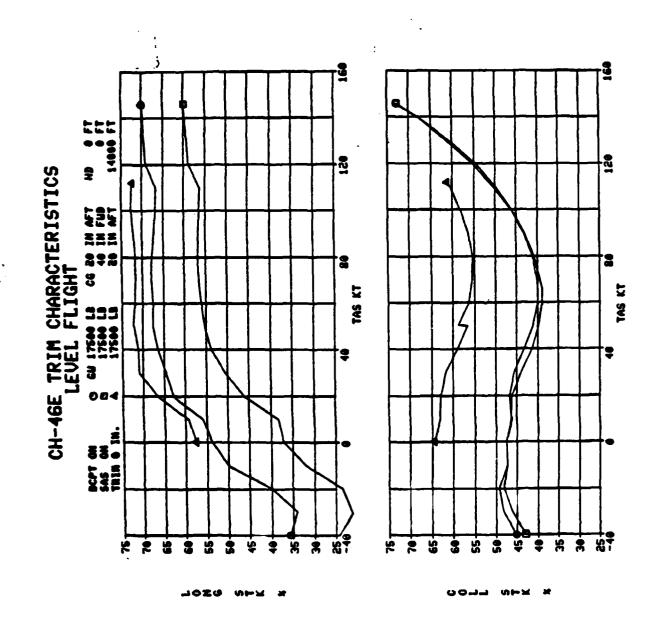


FIGURE 2A

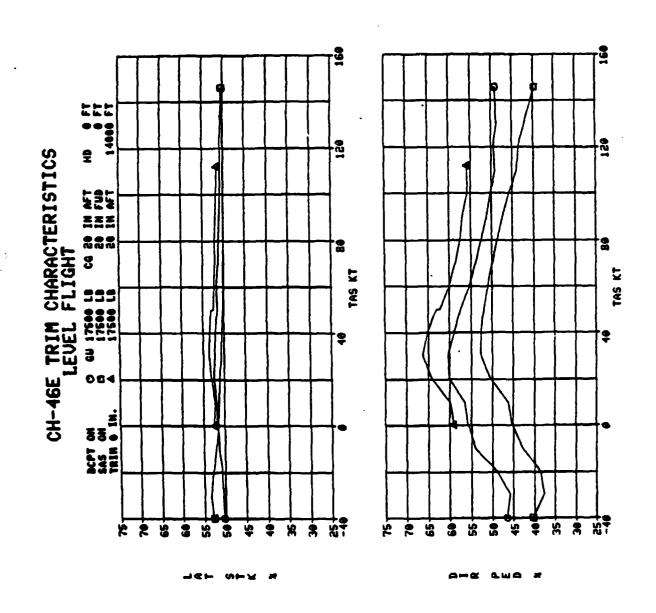
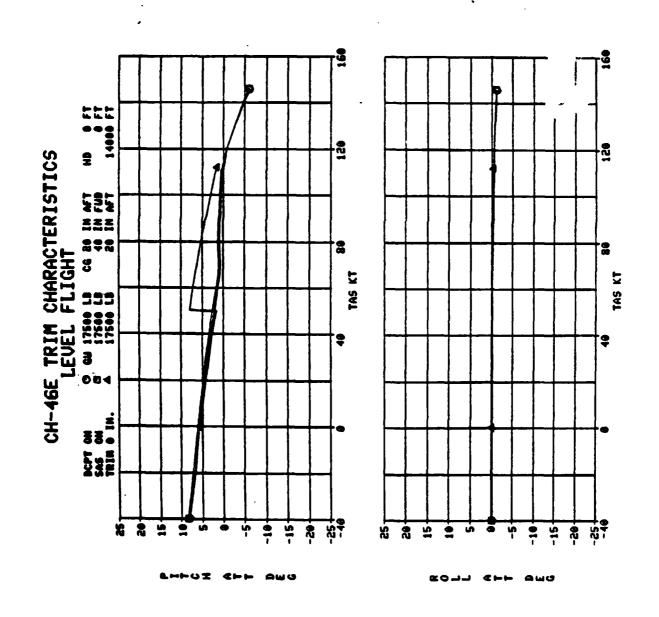


FIGURE 2B



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FIGURE 2C

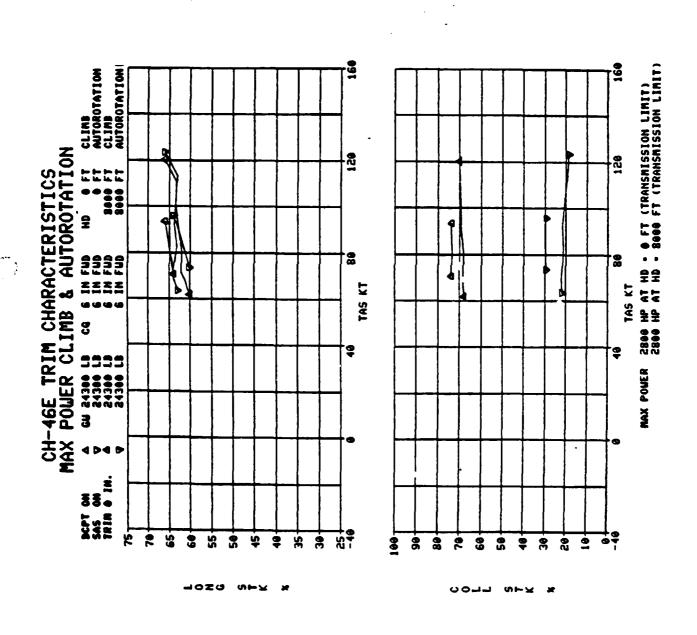


FIGURE 3A

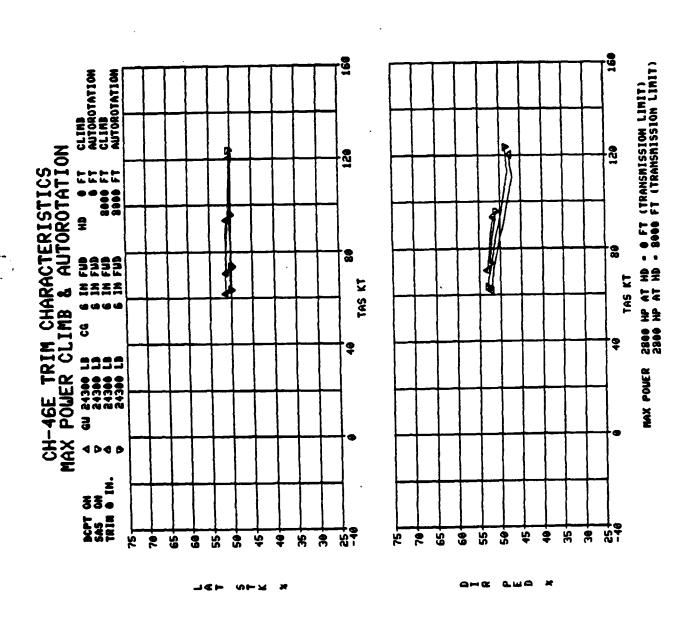
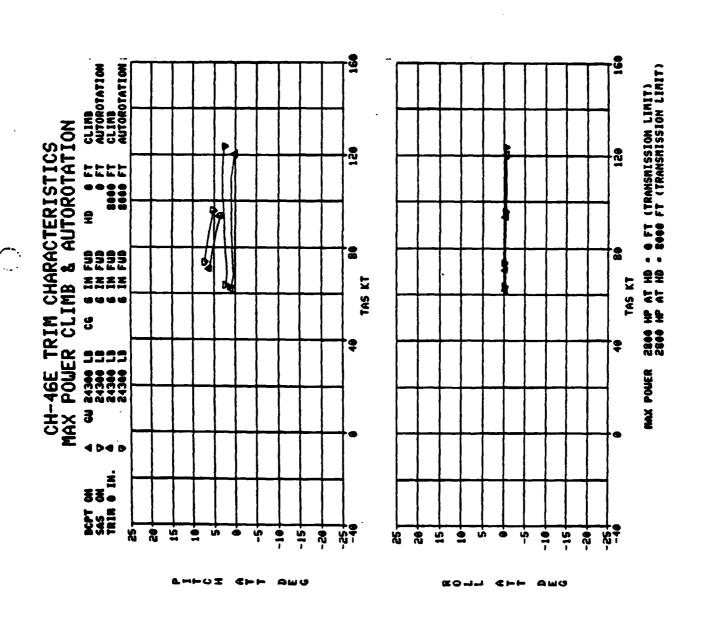


FIGURE 3B



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FIGURE 3C

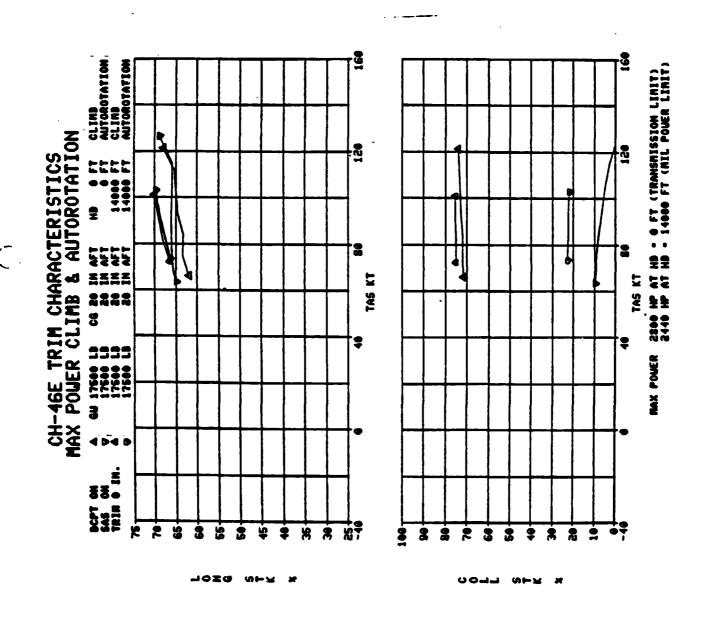


FIGURE 4A

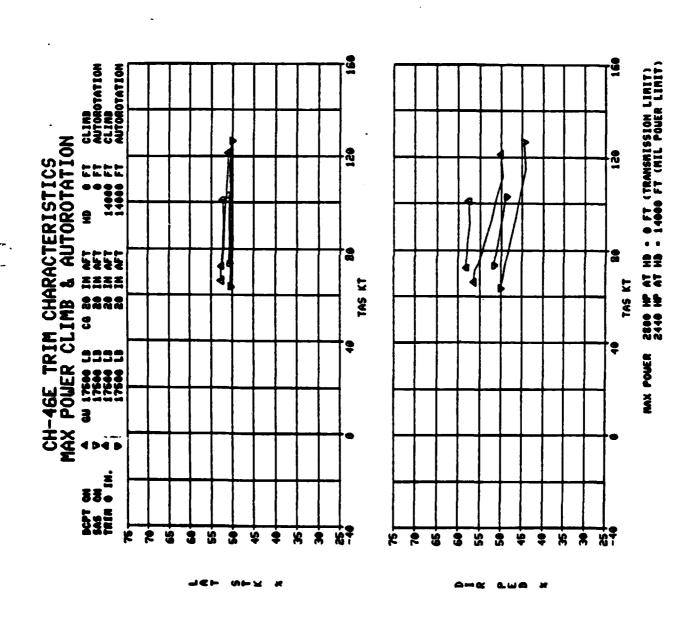


FIGURE 48

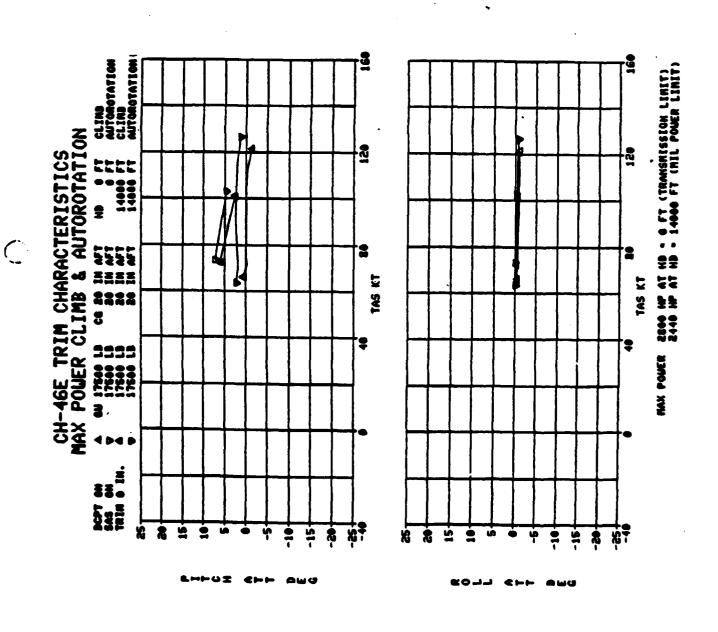


FIGURE 4C

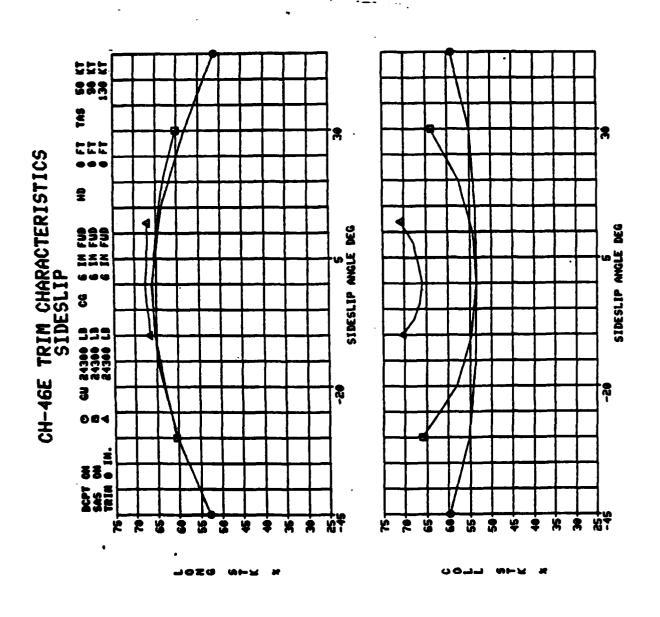


FIGURE 5A

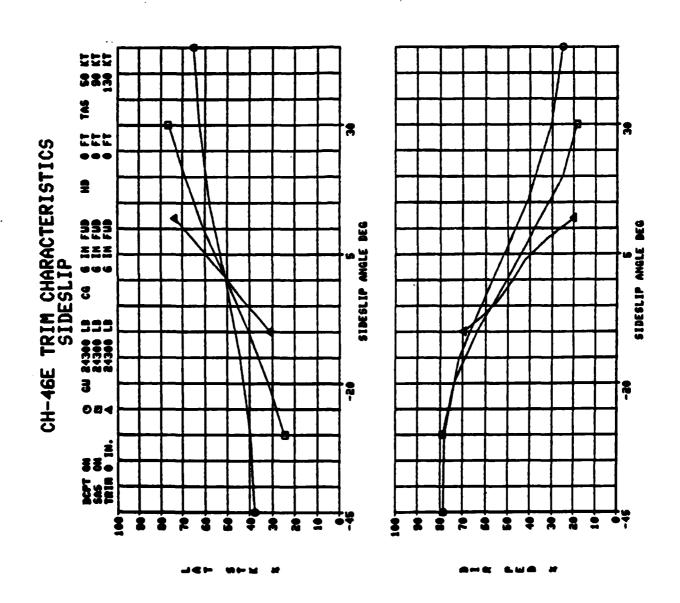


FIGURE 5B

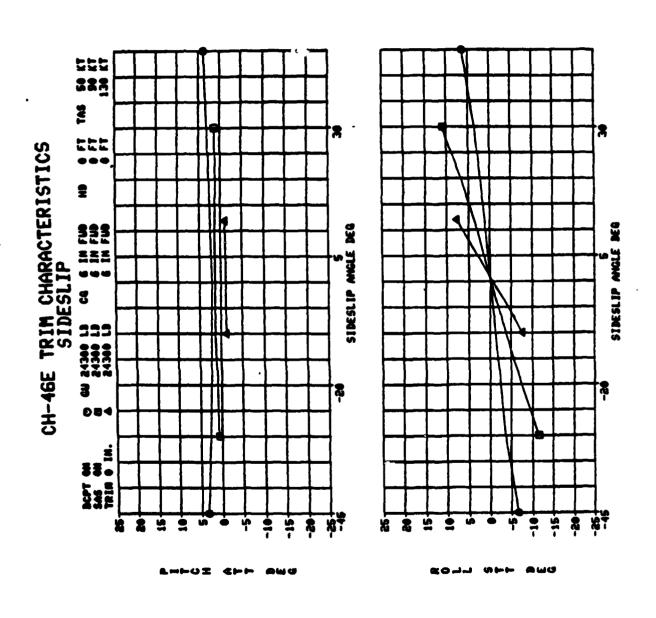


FIGURE 5C

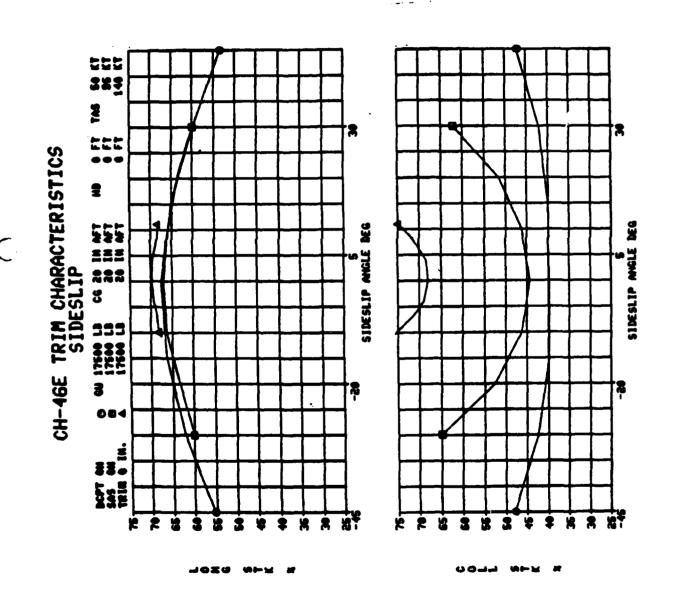


FIGURE 6A

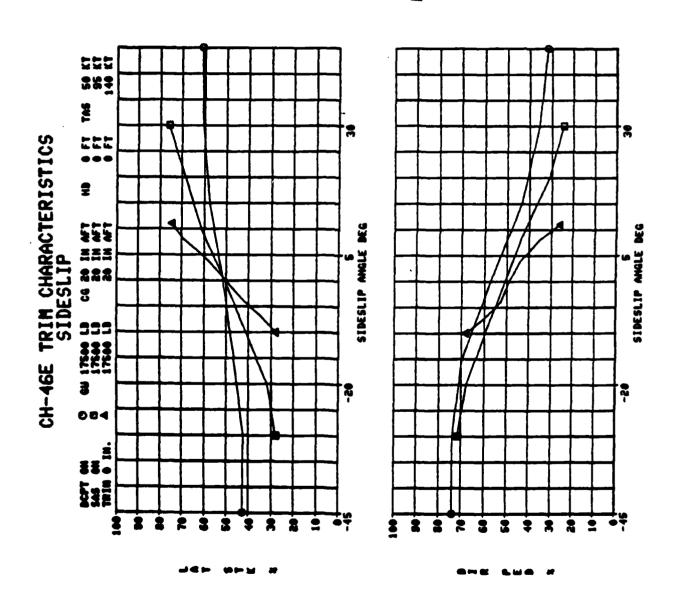


FIGURE 6B

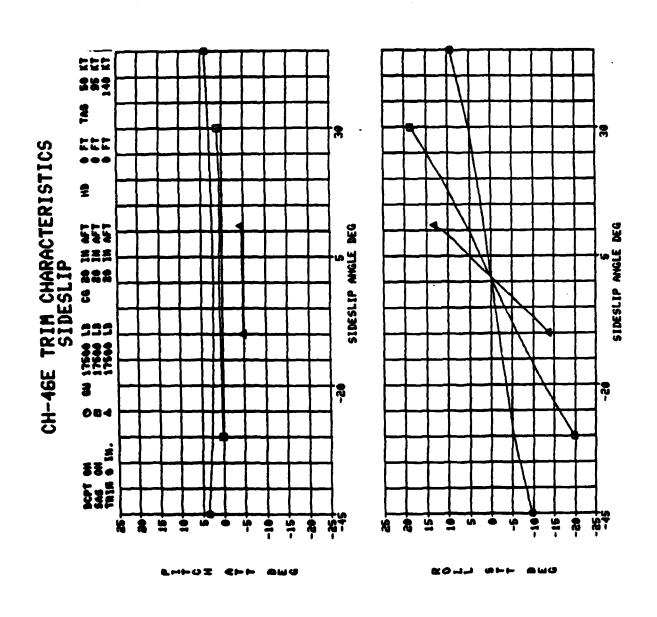


FIGURE 6C

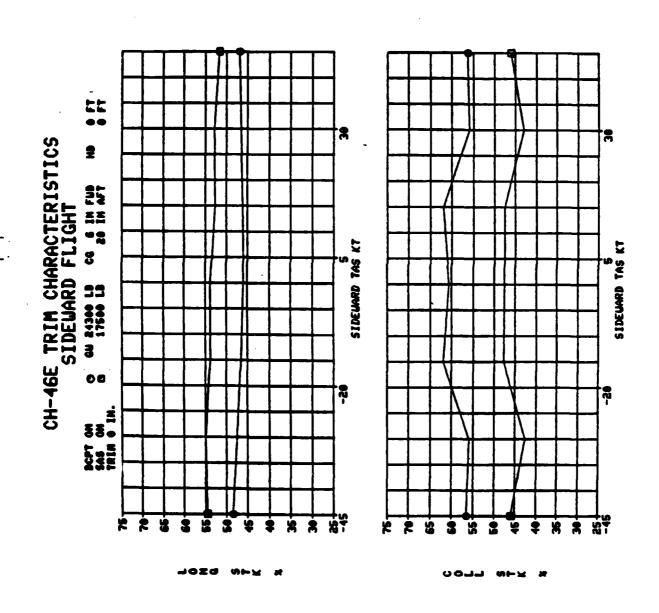


FIGURE 7A

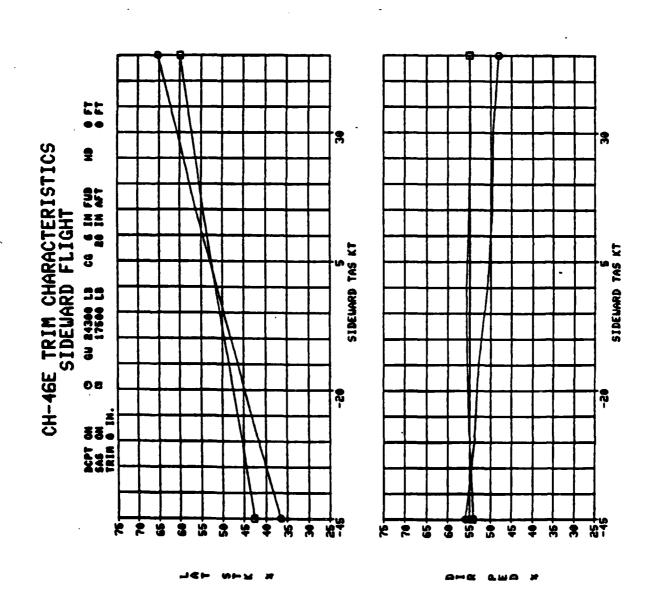


FIGURE 7B

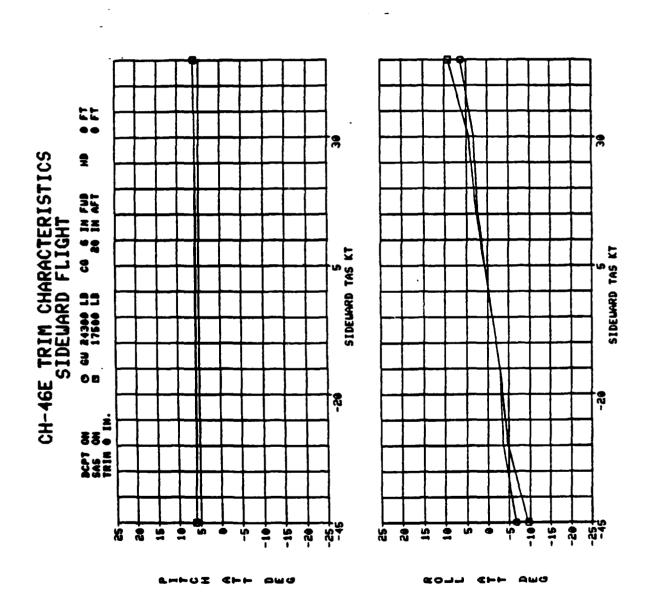


FIGURE 7C

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